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EVALUATION OF THE ACCURACY
OF A
RELIABILITY MEASUREMENT PROCEDURE
USING SIMULATION TECHNIQUES

by

Kenneth Alan Huffman



# UNITED STATES NAVAL POSTGRADUATE SCHOOL



# **THESIS**

EVALUATION OF THE ACCURACY
OF A
RELIABILITY MEASUREMENT PROCEDURE
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bу

Kenneth Alan Huffman

December 1968

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OF A

RELIABILITY MEASUREMENT PROCEDURE
USING SIMULATION TECHNIQUES

bу

Kenneth Alan Huffman Lieutenant, United States Navy B.A., Miami University, 1963

Submitted in partial fulfillment of the requirements for the degree of

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#### ABSTRACT

This study evaluates the accuracy of an established reliability measurement procedure (NAVWEPS OD 29304) by computer simulation. The reliability measurement procedure assumes components fail according to an Exponential Failure Law. This study tests the accuracy of that procedure when components obey a Weibull Failure Law or a Log Normal Failure Law.

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#### CHAPTER I

#### INTRODUCTION

The highly complex weapon systems being developed for the Navy today require a rapid assessment of system and subsystem reliability during research and development, and production phases. Currently there exists a statistical model, the <u>Guide Manual For Reliability Measurement Program</u> (NAVWEPS OD 29304), that "...can be utilized by all contractors for subsystem reliability measurement and by the Navy for weapon system reliability measurement." This model is versatile in that it permits the combination of test data from all levels and can be continuously updated as new test data becomes available. The model is thus a rapid approximation procedure for determining system reliability. However, the procedure has some restrictive assumptions which must be kept in mind.

One notable restrictive assumption is that all components are assumed to have a constant failure rate. This may not be necessarily true, although many times is a good approximation. This leads to the commonly used Exponential Failure Law. However, suppose the failures were actually given by a Weibull or a Log Normal Failure Law. Then this procedure must adequately estimate the system or subsystem

<sup>1</sup> Guide Manual For Reliability Measurement Program (NAVWEPS OD 29304, 15 May 1965), p. 3-1.

reliability under the Exponential Law assumption.

The purpose of this study is to evaluate the accuracy of the procedure in NAVWEPS OD 29304 when failures actually occur in a Weibull and Log Normal fashion.

The approximation procedure uses statistical estimates of failure rates based on sample data and are thus subject to statistical uncertainty. Therefore, at best, the procedure must yield a lower confidence limit on system or subsystem reliability, or an upper limit on failure rate. It is desired to test the accuracy of this procedure by simulating the distribution of the lower confidence limit for various systems and comparing the simulation results with the true reliability.

The method of simulation involves obtaining random Weibull and Log Normal failure time variates for components in a given system with a true system reliability,  $R_s$ . These quantities represent the times to failure of components provided they are less than the duration of the test (i.e. the planned test time). Then, using the statistical procedure in the <u>Guide Manual For Reliability Measurement Program</u>, a  $100(1-\alpha)\%$  lower confidence limit,  $\hat{R}_{s,L(\alpha)}$ , for  $R_s$  is obtained. A large number of replications of this process are completed to construct a distribution of  $\hat{R}_{s,L(\alpha)}$ . If the procedure is indeed accurate, the  $(1-\alpha)^{th}$  percentile point of the constructed distribution will be  $R_s$ . That is, if  $\hat{R}_{s,L(\alpha)}$  is a true  $100(1-\alpha)\%$  lower confidence limit for

 $R_s$ , then  $P(R_s \ge \widehat{R}_{s,L(\alpha)}) = 1-\alpha$ . If the  $(1-\alpha)^{th}$  percentile point of the distribution of  $\widehat{R}_{s,L(\alpha)}$  is denoted by  $A_{1-\alpha}$ , then  $A_{1-\alpha}$  should equal  $R_s$ , regardless of the set of parameter values chosen. Thus a measure of the accuracy of the procedure is given by the quantity  $\begin{vmatrix} A_{1-\alpha} - R_s \end{vmatrix}$ . Two other measures of the accuracy of this procedure are the mean and standard deviation of the distribution of  $\widehat{R}_{s,L(\alpha)}$ .

The accuracy of this procedure is examined for a variety of combinations of components and sets of parameter values. In all cases the systems under consideration consist of four components in series. All components are assumed to behave independently so the system reliability is the product of the four component reliabilities.

As expected, the closer the Weibull and Log Normal distributions approximate the Exponential, the better the accuracy of the procedure. However, for some components with failure time distributions differing considerably from the Exponential, the accuracy is fairly good using this method of reliability measurement. If the absolute difference  $\begin{vmatrix} A_{1-\alpha} - R_{s} \end{vmatrix}$  is less than .02, the accuracy is called good. The accuracy of the procedure is good for planned test times of short duration and diminishes as they are increased. The study concludes that the reliability measurement procedure, being an approximation procedure, is useful for some systems following Weibull and Log Normal Failure Laws provided they do not deviate from the Exponential Failure Law by too significant an amount.

#### CHAPTER II

## STATISTICAL RELIABILITY MODEL<sup>2</sup>

The statistical Reliability Model provides the guide lines for reliability measurement on Navy weapons systems. The "Guide Manual" can be used by both contractors for subsystem reliability measurement and by the Navy for weapon system reliability measurement. A brief description of this model's assumptions and methodology are presented in this chapter.

The system reliability obtained is based upon testing components for a time duration, called the planned test time, or until failure occurs under a given stress condition.

Also, the model provides for testing a system for component failure by the same method.

This model considers components in a complex system which must operate successfully over a defined "mission", lasting a specified duration of time. Time can thus be measured in "mission units". System reliability is thus a function of environmental and usage stresses which include vibration, shock, etc., and the operating vs. non-operating conditions as well as a function of the associated part failure rate parameters together with the time duration of the environments and usage stresses. The reliability

<sup>&</sup>lt;sup>2</sup>Guide Manual For Reliability Measurement Program (NAVWEPS OD 29304, 15 May 1965), pp. 3-1 to 3-15.

measurement system is dependent upon the following assumptions:

- 1. Constant failure rate. The exponential failure law is assumed to hold, mainly for its common usage, mathematical simplicity, practical simplicity, and reasonability.
- 2. Additivity of stress effects. "The failure rate induced by two simultaneously acting stresses is equal to the sum of the failure rates due to the two stresses acting sequentially."3 Thus, mission experience can be simulated by adding data from separate environmental tests.
- 3. Independence of component failures. "...This is assumed because components are normally tested individually by type, and subsystem reliability is estimated using component and other applicable test results."
- 4. Failure rate constancy. "The failure rate is considered a function of only the stress acting." In other words, the component being tested has no "memory" as to previous stresses of a different type.

Computation of the system reliability is accomplished by the following procedure:

An unbiased failure rate estimate is computed, providing the failure rates are anticipated as being small. For ease of computation, a simplified unbiased estimator, which is slightly conservative, is used. For the i<sup>th</sup> component, the unbiased estimator,  $\hat{\lambda}_i$ , is given by:

<sup>3&</sup>lt;sub>Ibid</sub>. p.3-3.

<sup>4</sup>Ibid.

<sup>5</sup> Ibid.

$$\hat{\lambda}_{i} = \frac{f_{i}}{N_{i}} \cdot \frac{2N_{i}}{2N_{i} + 1}$$

$$\sum_{j=1}^{\Sigma} T_{ij} \cdot 2N_{i} + 1$$

where  $N_i$  = the sample size of the i<sup>th</sup> component (i.e. the number of components of type i tested).

 $T_{i}$  = the test time (time to failure) of the component of type i.

 $T_{oi}$  = the planned test time for component i. and  $f_{i}$  = the number of  $N_{i}$  failure times which are less than  $T_{oi}$ .

The system failure rate estimate is then given by:

$$\hat{\lambda} = \sum_{i=1}^{m} \hat{\lambda}_{i}$$
 (2)

where m is the number of components in the system. The component number assumes values i = 1, ..., m.

The variance of the unbiased failure rate estimate is given by:

Variance = 
$$\sum_{i=1}^{m} (\hat{\lambda}_i/S_i)$$

where

$$S_i = \sum_{j=1}^{N_i} T_{ij} =$$
the sum of all test times

accumulated on the N components of type i.

These estimates of failure rates for components in a given test condition can be continually updated during the development phase as more data becomes available. Also, during any higher level of assemblage, further testing can also yield data to modify the failure rate estimate. Data is obtained indicating the time that each component operates during the system test.

The reliability estimate for the system is then given by  $\hat{R}_s = e^{-\hat{\lambda}}$  where  $\hat{\lambda}$  is the appropriate failure rate estimate and only series systems or subsystems are considered, given by formula (2).

The reliability estimates obtained are subject to statistical uncertainty, therefore the important item to observe is the confidence interval about the estimate. Here, the lower confidence limit on system reliability becomes the pertinent quantity to observe. The "Guide Manual" bases the lower confidence limit on Normal theory and it is corrected to compensate for small values of  $\lambda$ . The upper limit on failure rate,  $\hat{\lambda}_u$ , provides the corresponding lower confidence limit on reliability by  $\hat{R}_{s,L} = e^{-\hat{\lambda}_u}$ . The upper limit on failure rate is given by:

$$\hat{\lambda}_{u} = \begin{cases} \frac{2\hat{\lambda} + K^{2}\hat{c} + \sqrt{4\hat{\lambda}K^{2}\hat{c} + K^{4}\hat{c}^{2}}}{2} & \text{for } \sum_{i=1}^{m} f_{i} > 0 \\ \frac{K^{2}}{n} \sum_{i=1}^{m} (1/S_{i}) & \text{for } \sum_{i=1}^{m} f_{i} = 0 \end{cases}$$
(3)

where

$$\hat{c} = \frac{\sum_{i=1}^{m} (\hat{\lambda}_{i}/s_{i})}{\hat{\lambda}}$$

n = the number of component-environment-test condition terms in the summation, and K = the percentage point of the Normal Distribution. The values of K for a given confidence level are not completely appropriate for small values of  $\lambda$ . Consequently, a correction factor,  $\beta$ , is used to obtain desired precision. Beta values are tabulated for the 80% confidence limit in reference (2). Thus the percentage point of the Normal Distribution, K, is modified by  $\beta$ K and replaced in formula (3).

A more detailed description of this reliability measurement method plus examples of the procedure are given in reference (2).

#### CHAPTER III

#### SIMULATION PROCEDURE

Suppose the assumption of constant failure rate used in the previously described statistical model is relaxed. It is now desired to test the accuracy of that model when failures occur according to a Weibull Failure Law and a Log Normal Failure Law. For testing purposes by simulation techniques, consider a series system with four components, each component having a well defined failure law.

The true reliability for each component of this series system is given by  $R_i$ , i = 1,2,3,4, and thus the true system reliability is

$$R_{S} = \sum_{i=1}^{L} R_{i} . \tag{4}$$

Using the statistical model, an estimate of the  $(1-\alpha)^{th}$  percentile confidence limit of the system,  $\widehat{R}_{s,L(\alpha)}$ , can be obtained by computer simulation. This is a random variable as calculated from the statistical model. If

P( 
$$R_s \ge \hat{R}_{s,L(\alpha)}$$
) = 1-\alpha

then, in fact,  $\hat{R}_{s,L(\alpha)}$  is an exact  $(1-\alpha)^{th}$  percentile lower confidence limit for  $R_s$ . This says that  $R_s$  is always the  $(1-\alpha)^{th}$  percentile point of the probability distribution of  $\hat{R}_{s,L(\alpha)}$ . The distribution of  $\hat{R}_{s,L(\alpha)}$  is constructed by

computer simulation. Letting  $A_{1-\alpha}$  be the  $(1-\alpha)^{th}$  percentile point of the distribution of  $\widehat{R}_{s,L(\alpha)}$ , then the absolute difference,  $\left|A_{1-\alpha}-R_{s}\right|$ , is a measure of the accuracy of the statistical model.

Two other measures of the accuracy of this model are the estimated mean,  $\widehat{R}_{s,L(\alpha)}$ , and the estimated standard deviation,  $s \widehat{R}_{s,L(\alpha)}$ , which give some assurance that the accuracy of this model are the estimated standard deviation,  $\widehat{R}_{s,L(\alpha)}$  actual values generated by the procedure are reasonable.

The simulation procedure itself is now discussed for the series system of four components:

The distribution of  $\widehat{R}_{s,L(\alpha)}$  is constructed by generating 500 random observations on  $\widehat{R}_{s,L(\alpha)}$  for a given system, number of components tested, and reliability of the components. For this study a four component series system is used, with each component having reliability  $R_i$ , i=1,2,3,4. The reliability,  $R_i$ , is defined for each component by  $R_i=R_i(1)=P(T_i>1)$ , where  $T_i$  is the time to failure random variable for a component of type i. This establishes the parameter values for a particular failure rate distribution. For example, if a component is to obey a Weibull Failure Law,

$$R(t) = e^{-(\lambda t)^{\beta}}$$
 (5)

and R(1) =  $e^{-\lambda^{\beta}}$  = .995, then, for  $\beta$  = 2, this implies that  $\lambda$  = .0707.

Similarly, for the Log Normal Failure Law, if

$$R(t) = e^{Z} = 1 - \Phi \left[ \frac{\ln t - \mu}{\sigma} \right]$$
 (6)

and  $R(1) = 1 - \Phi(-m/\sigma) = .995$ , then, for  $\sigma = 1$ , this implies that m = 2.576. The failure rate functions determined from the reliabilities and selected parameters are displayed in Appendix I.

A random number generator is used to obtain a sample of N<sub>i</sub> tested components of type i which fail according to their respective failure rate distributions. This Monte Carlo method draws a random number which is Uniformly Distributed (0,1) and converts it to a Weibull variate or a Log Normal variate to obtain the time to failure of the i<sup>th</sup> component, T<sub>i</sub>. For the Weibull Distribution,  $e^{-(\lambda T)^{\beta}} = Y$ , where Y is distributed as Uniform (0,1), implies that

$$T = \frac{(-\ln Y)^{1/\beta}}{\lambda} \tag{7}$$

is a random Weibull variate. For the Log Normal Distribution, if X is a random variable distributed as Normal(0,1), then  $Z = \sigma X + \mu$  is a random variable distributed as Normal  $(\mu, \sigma^2)$  and thus  $T = e^Z$  is a random Log Normal variate. In this manner,  $N_i$  failure times are generated for each of the four components:

$$T_{11}$$
  $T_{12}$   $T_{13}$  . . .  $T_{1N_1}$ 
 $T_{21}$   $T_{22}$   $T_{23}$  . . .  $T_{2N_2}$ 
 $T_{31}$   $T_{32}$   $T_{33}$  . . .  $T_{3N_3}$ 
 $T_{41}$   $T_{42}$   $T_{43}$  . . .  $T_{4N_{11}}$ 

Using these failure times in the statistical procedure discussed in the previous chapter, values of  $\hat{R}_s$  and  $\hat{R}_{s,L(\alpha)}$  are obtained with that model. Each  $T_{ij}$  is examined for failure before the end of the planned test time or else terminated at the end of the planned test time. This same procedure is completed 500 times to generate 500 random observations, thus constructing the distributions of  $\hat{R}_s$  and  $\hat{R}_{s,L(\alpha)}$ .

The mean and standard deviation of the distributions are determined by:

$$\overline{\hat{R}}_{s,L(\alpha)} = \frac{1}{500} \sum_{k=1}^{500} \hat{R}_{s,L(\alpha)_k}$$
 (8)

$$\vec{\hat{R}}_{s} = \frac{1}{500} \quad \sum_{k=1}^{500} \quad \hat{\hat{R}}_{s_{k}}$$
(9)

$$s_{\hat{R}_{s,L(\alpha)}} = \sqrt{\frac{1}{500}} \sum_{k=1}^{500} (\hat{R}_{s,L(\alpha)_{k}} - \overline{\hat{R}}_{s,L(\alpha)})^{2}$$
 (10)

$$s_{\hat{R}_{s}} = \sqrt{\frac{1}{500}} \frac{500}{\Sigma} (\hat{R}_{s_{k}} - \bar{R}_{s})^{2}$$
 (11)

The 500 values of  $\hat{R}_{s,L(\alpha)}$  are then ordered and the  $(1-\alpha)^{th}$  percentile point of the distribution is obtained. Thus, the value  $A_{1-\alpha}$  is available for comparison with  $R_s$ . This process is done merely by counting down to the  $101^{st}$  value of  $\hat{R}_{s,L(\alpha)}$  from the largest, which represents the  $80^{th}$  percentile point.

Several cases are examined with various failure rate distributions and with various combinations of components. For each case examined, the planned test time,  $T_{oi}$ , varies to produce an optimistic or pessimistic reliability test. If  $T_{oi}$  = .5, the system will yield an optimistic test since the planned test time will be reached before most components fail. The opposite will occur when  $T_{oi}$  = 5. A  $T_{oi}$  chosen such that the average failure rate is equal to the failure rate at time one appears to be a good test of system reliability for this procedure. That is, make  $T_{oi}$  such that it satisfies the equation

$$\frac{1}{T_{0i}} \int_{0}^{T_{0i}} z_{i}(t) dt = z_{i}(1)$$
 (12)

where z(t) is the failure rate function; the failure distribution divided by the reliability function: z(t) = f(t)/R(t).

All cases studied by simulation procedures are given in Table I. All simulation results of these cases are given in Table II.

TABLE I

CASES STUDIED BY SIMULATION PROCEDURES

Case No.	Compo- nent No.	Failure Law	Parameters	Ri	Rs
1	1	Weibull	β=1.5, λ=.0293	.995	.98
	2	Weibull	β=2.0, λ=.0707	•995	
	3	Log Normal	$\sigma=1.0, m=2.576$	.995	
	4	Log Normal	$\sigma = 2.0, \mu = 5.152$	•995	
2	1	Weibull	β=1.5, λ=.0293	•995	.98
	2	Weibull	$\beta=1.5, \lambda=.0293$	•995	
	3	Log Normal	$G=2.0, \mu=5.152$	•995	
	4	Log Normal	σ=2.0, μ=5.152	•995	
3	ı	Weibull	β=1.33, λ=.0378	.987	•95
	2	Weibull	$\beta = 1.33, \lambda = .0378$	.987	
	3	Log Normal	σ=1.5, μ=3.345	.987	
	4	Log Normal	σ=1.5, μ=3.345	.987	
4	1	weibull	β=1.2, λ=.0485	.974	.90
	2	Weibull	β=1.2, λ=.0485	.974	
	3	Log Normal	σ=1.5, μ=2.919	.974	
	4	Log Normal	σ=1.5,μ=2.919	.974	
5	1	Weibull	β=1.2, λ=.0485	.974	.90
	2	Weibull	$\beta=1.2, \lambda=.0485$	.974	
	3	Weibull	$\beta = 1.2, \lambda = .0485$	.974	
	4	Weibull	$\beta=1.2, \lambda=.0485$	.974	

TABLE I (Continued)

Case No.	Compo- nent No.	Failure Law	Parameters	R	Rs
6	1	Log Normal	$\sigma=1.5, \mu=2.919$	.974	.90
	2	Log Normal	$\sigma=1.5, \mu=2.919$	.974	
	3	Log Normal	$\sigma=1.5, \mu=2.919$	.974	
	4	Log Normal	σ=1.5, μ=2.919	.974	
7	1	Weibull	β=1.1, λ=.0483	.965	.867
	2	Weibull	$\beta$ =1.1, $\lambda$ =.0483	.965	
	3	Weibull	$\beta = 1.1, \lambda = .0483$	.965	
	4	Weibull	β=1.1, λ=.0483	.965	
8	1	Log Normal	$\sigma = 2.0, \mu = 4.0$	.977	.912
	2	Log Normal	$\sigma = 2.0, \mu = 4.0$	.977	
	3	Log Normal	$\sigma=2.0, \mu=4.0$	•977	
	4	Log Normal	σ=2.0, μ=4.0	.977	

#### CHAPTER IV

#### RESULTS AND CONCLUSIONS

In the preceding chapters the statistical procedure from the <u>Guide Manual For Reliability Measurement Program</u> has been briefly explained to obtain a  $100(1-\alpha)\%$  lower confidence limit on system reliability. The simulation of this method on a computer has been explained and the technique for measuring the accuracy of the statistical procedure stated. Now, from the results of the simulation, some conclusions may be drawn.

The accuracy of the statistical procedure has been examined for eight cases which represent different combinations of components for various sets of parameter values. The results for these cases are given in Table II. The mean and standard deviation of the 500 values of  $\hat{R}_{s,L(\alpha)}$  and  $\hat{R}_{s}$  are given, along with the  $(1-\alpha)^{th}$  percentile point,  $A_{1-\alpha}$ . In all cases studied,  $\alpha$  was taken to be .20.

The accuracy of the procedure is measured by the absolute difference between the  $80^{\rm th}$  percentile of the distribution of  ${\rm \hat{R}_{s,L(.20)}}$ , which is  ${\rm A_{.80}}$ , and the actual system reliability,  ${\rm R_{s}}$  (i.e. accuracy =  ${\rm A_{.80}}$  -  ${\rm R_{s}}$ ).

Also stated in the results, is the quantity TT, which is the amount of testing relative to the component unreliabilities. In other words, this is a measure of the amount of testing required to achieve a desired accuracy for a given system with component failure rates,  $z_i(1)$ ,

and planned test times, Toi. Then TT is given by

$$TT = \sum_{i=1}^{m} N_i z_i(1) T_{0i} . \qquad (13)$$

As an example of the accuracy study, consider Case 1, where the system consists of four components in series displaying the following failure laws:

Component 1: Weibull,  $\beta=1.5$ ,  $\lambda=.0293$ Component 2: Weibull,  $\beta=2.0$ ,  $\lambda=.0707$ Component 3: Log Normal,  $\sigma=1.0$ ,  $\mu=2.576$ Component 4: Log Normal,  $\sigma=2.0$ ,  $\mu=5.152$ 

The component reliabilities,  $R_1$  = .995, for all four components, so the system reliability,  $R_S$  = .9801. The measure of accuracy for the optimistic planned test time,  $T_{oi}$  = .5, is  $\left|A_{.80} - R_{S}\right|$  = .008 for 500 components of each type tested. The pessimistic planned test time,  $T_{oi}$  = 5.0, yields an accuracy of .055 for 100 components of each type tested. The <u>ad hoc</u> planned test times,  $T_{ol}$  = 2.25,  $T_{o2}$  = 2.0,  $T_{o3}$  = 1.95, and  $T_{o4}$  = 3.25, as determined from formula (12), give an accuracy of .030 for  $N_i$  = 100. The accuracy for the <u>ad hoc</u> planned test times is only fair. This planned test time represents the best measure of accuracy of the

It can be seen from Table II and Appendix I that the accuracy of the procedure is a function of how closely the Weibull and Log Normal failure rate functions approximate a constant failure rate function. The greater the Weibull

and Log Normal failure rate functions deviate from the constant failure rate, the more inaccurate the procedure. Also, the accuracy decreases as the planned test time is increased. The Weibull failure rate function increases in time and the Log Normal failure rate function increases then decreases in time, thus deviating from the constant failure rate for large planned test times.

It can thus be concluded from the study that the statistical procedure can be useful for reliability approximations of a system, if the components that fail according to these non-constant rates do not deviate too significantly from a constant failure rate.

TABLE II SIMULATION RESULTS CASE 1

( つ) コ * か
std.dev.
,0244
.0175
4200.
.0178
.0129
.0052
.0218
.0155
.0059
.0159
.0105
· 0084

TABLE II (Continued)

CASE 2	$\hat{A}_{S,L(\alpha)}$	mean std.dev.	.9248 .0255 .9905	.9451 .0159 .9902	.9783 .0083	9696. 7510. 6646.	.9542 .0094 .9631	.95960038 .9629	.9524 .0178 4595	.9593 .0125 .9713	1179. 7400. 0796.
	•	$T_{\text{O1}}$ $N_{1}$ mean		100 .945		646. 05 0.5	•		= 1 50 .	i = 3 100 .959	500

TABLE II (Continued)

		TT	1.3	2.5	12.7	12.7	25.4	127.	2.6	5.5	25.8
		accuracy	.012	.018	0.014	:043	.041	040°	.028	.027	.019
.95		A.80	9326°	.9683	.9643	0206°	0606.	.9101	.9220	.9235	.9312
E S	K S	std.dev.	0460°	.0249	.0109	.0177	.0131	.0055	.0229	.0168	0000°
CASE 3		mean	.9701	.9689	.9681	6806°	4806°	7806°	.9262	.9245	.9242
Ö	$\hat{\mathbb{R}}_{s,L(\alpha)}$	std.dev.	2140°	,0304	.0123	.0188	.0138	. 0057	,0254	.0180	.0073
	RS S	mean	0006.	.9287	.9552	.8903	. 8969	4606.	.9002	4206.	.9178
		N,	95	100	200	50	100	200	50	100	500
		Toi	.5			5.0			.37, 1=	2.72	- ( ~ ) ·

TABLE II (Continued)

		$_{ m TT}$	2.6	5.2	26.0	26.0	52.0	260.	16,1	32.2	161.	
		accuracy	.012	.018	.026	940°	540°	940°	.038	.036	.031	
06.		A.80	.8876	.9179	.9257	.8537	.8545	.8542	.8617	0498.	.8693	
ഷ യ 	R S	std.dev.	.0536	.0379	.0160	4220.	.0157	6900°	.0267	4610°	.0082	
CASE 4	\H	mean	.9303	.9279	.9279	.8542	4458.	.8540	.8655	0498.	.8638	
์ บ	Α̈́s, L(α)	std.dev.	.0620	.0432	.0171	.0232	.0163	.0070	.0287	.0204	7800.	
	KH S	mean	.8528	.8805	0116.	.8339	.8405	6248°	.8375	8948.	.8563	1
		N,	50	100	200	90	100	200	95	100	200	
		$T_{01}$	5.			5.0			49, 1=	3.70, 1 = 3	 	

(Continued) TABLE II

.90

11

3

CASE

5.2 10.4 12.9 25.9 26.0 52.0 52.0 78.0 129. 104. 156. 104° TT 208. accuracy .005 .035 .014 .035 .068 .023 033 .052 690. .052 .062 .021 .061 A.80 .8865 8448 .8485 .8954 .8789 .8633 .8655 .8379 .8324 .8314 .8772 .8387 .9347 std.dev. ഷ 9040° .0215 .0282 .0280 ·0194 0600° .0148 .0163 0118 00100 .0129 .0092 .0140 S S .8823 .8818 .8808 .8658 9848. 9848° .8325 .9007 .8999 .8657 .8389 .8389 .8321 mean std.dev. 9410. .0103 .0452 0600. .0170 .0134 .0307 .0302 .0197 .0154 .0121 .0221 .0097 Rs,L(a) .8298 .8204 .8237 7648° .8670 .8627 .8730 .8459 .8522 .8338 .8383 .8259 mean .8517 50 100 500 50 100 50 100 50 100 50 100 50 100 Z Z 2.49 Toi 10 15 20 5

TABLE II (Continued)

		TT	2.6	5.5	26.0	5.2	10.4	52.0	19.2	38.4	192.
		accuracy	.038	240.	.038	° 005	.013	000%	450°	450°	.053
06.		A.80	.9376	9176.	.9383	.8952	.9131	.9041	.8458	4948°	.8470
R S =	\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	std.dev.	5940.	.0317	.0141	6140°	.0295	.0129	.0260	.0181	.0087
CASE 6		mean	0946°	2446.	.9428	.9051	9606.	.9029	.8478	.8471	.8471
O	Âs, L(α)	std.dev.	6450°	.0370	.0153	.0468	.0322	.0135	4920°	.0188	. 0089
	<¤,	mean	.8712	.8992	.9271	9458°	.8712	4168.	.8227	.8307	.8400
		N	50	100	200	50	100	500	50	100	500
		Toi	5.			J.O			3.70		

TABLE II (Continued)

.867		A,80 accuracy Tr	.8736 .006 7.0	.9234 .056 14.0	.8461 .021 35.0	.8464 .021 70.0	.8370 .030 70.0	.8371 .030 140.	.8323 .035 105.	.8323 .035 210.	.8295 .038 140.	.8283 .039 280.
R <sub>S</sub> ≡ °S	R <sub>S</sub>	std.dev. A	. 0462	6. 6160.	.0228	. 0159	8. 1710.	.0122	. 0146	8. 4010.	.0134	8. 5600.
CASE 7		mean	.8678	.8675	6948°	2948°	.8376	.8377	.8328	.8326	.8293	.8290
	$\hat{R}_{s,L(\alpha)}$	std.dev.	.0502	.0340	.0237	.0165	.0178	.0126	.0151	2010.	0410.	8600.
	<¤"	mean	.8132	.8321	.8263	.8325	.8222	.8270	.8195	. 8233	.8170	.8205
		N1	50	100	50	100	50	100	50	100	50	100
		Toi	ч		2		10		1.5		20	

TABLE II (Continued)

CASE 8  $R_{S} = .912$ 

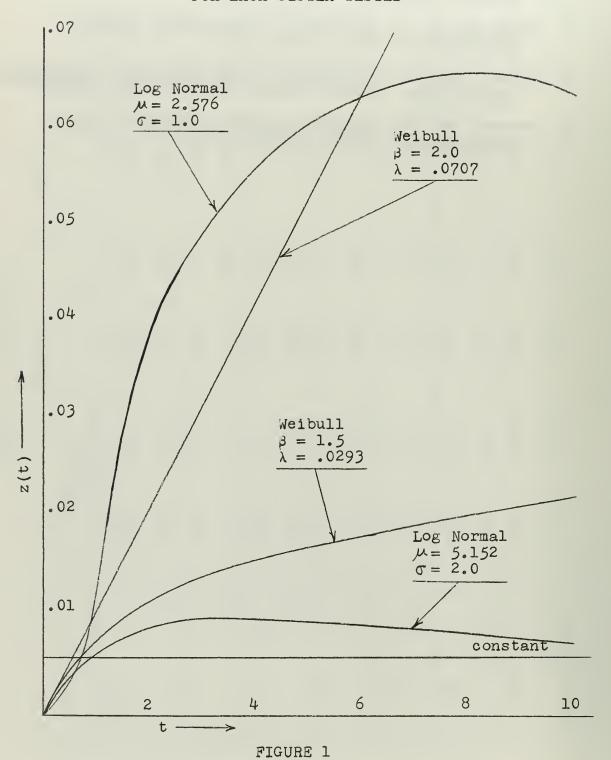
lev.       mean       std.dev.       A.80         51       .9166       .0399       .8959         51       .9166       .0399       .8959         52       .9152       .0284       .9348         53       .9028       .0284       .9348         59       .9028       .0210       .9075         59       .9044       .0185       .9013         50       .9044       .0185       .9158         59       .9138       .0135       .9158         59       .9135       .0095       .9145         60       .9211       .0111       .9228         70       .9267       .0096       .9279         70       .9264       .0069       .9273         70       .9566       .0047       .9577         70       .9567       .0033       .9571			Rs	βs, L(α)		R S		Annichaean administrator e delinio de producto de composto de composto de la composta del la composta de la composta del la composta de la composta del la composta de la composta de la composta de la composta de la c	
50         .8674         .0451         .9166         .0399         .8959           100         .8839         .0313         .9152         .0284         .9348           100         .8839         .0313         .9152         .0284         .9348           50         .8707         .0329         .9035         .0300         .9006           100         .8813         .0223         .9028         .0210         .9075           100         .8926         .0130         .9044         .0185         .9013           100         .8926         .0130         .9043         .0124         .9034           100         .9054         .0089         .9138         .0136         .9148           100         .9143         .0183         .9208         .9148         .9148           100         .9143         .0083         .9264         .0096         .9279           100         .9209         .0072         .9264         .0069         .9279           100         .9209         .0050         .9567         .0069         .9279           20         .9543         .0050         .9567         .0033         .9571           100	Toi	N,	mean	std.dev.		std.dev.	A.80	accuracy	TT
100       .8839       .0313       .9152       .0284       .9348         50       .8707       .0329       .9035       .0300       .9006         100       .8813       .0223       .9028       .0210       .9075         50       .8853       .0195       .9044       .0185       .9013         100       .8926       .0130       .9043       .0124       .9034         100       .9024       .0143       .9138       .0135       .9158         100       .9054       .0099       .9135       .0035       .9145         50       .9118       .0118       .9218       .0019       .9218         100       .9143       .0083       .9208       .0079       .9218         100       .9143       .0083       .9208       .0079       .9218         100       .9209       .0072       .9264       .0069       .9273         50       .9532       .0050       .9566       .0047       .9577         100       .9543       .0034       .9587       .9033       .9571	П	50	4298.	.0451	9916.	.0399	. 8959	910.	9.4
50       .8707       .0329       .9035       .0300       .9006         100       .8813       .0223       .9028       .0210       .9075         50       .8853       .0195       .9044       .0185       .9013         100       .8926       .0130       .9043       .0124       .9034         50       .9021       .0143       .9138       .0136       .9158         100       .9054       .0099       .9135       .0095       .9145         50       .9118       .0118       .9211       .0111       .9228         100       .9143       .0083       .9208       .9059       .9216         50       .9187       .0101       .9264       .0069       .9279         100       .9532       .0050       .9566       .0047       .9577         100       .9543       .0034       .9567       .0033       .9571		100	.8839	.0313	.9152	.0284	.9348	.023	9.2
100       .8813       .0223       .9028       .0210       .9075         50       .8853       .0195       .9044       .0185       .9013         100       .8926       .0130       .9043       .0124       .9034         100       .8926       .0138       .0135       .9158         100       .9054       .0099       .9135       .0095       .9145         100       .9143       .0083       .9208       .0079       .9279         100       .9187       .0101       .9264       .0069       .9273         100       .9532       .0050       .9566       .0047       .9577         100       .9543       .0034       .9567       .0033       .9577	2	50	.8707	.0329	.9035	.0300	9006	.011	9.5
50         .8853         .0195         .9044         .0185         .9013           100         .8926         .0130         .9043         .0124         .9034           100         .9926         .0136         .9138         .0135         .9158           100         .9054         .0099         .9135         .0095         .9145           50         .9118         .0118         .9211         .0111         .9228           100         .9143         .0083         .9208         .0079         .9216           100         .9187         .0101         .9267         .0069         .9273           100         .9209         .0072         .9264         .0069         .9273           50         .9532         .0050         .9566         .0047         .9577           100         .9543         .0034         .9567         .0033         .9571		100	.8813	.0223	.9028	.0210	.9075	†00°	18.4
100       .8926       .0130       .9043       .0124       .9034         50       .9021       .0143       .9138       .0135       .9158         100       .9054       .0099       .9135       .0095       .9145         50       .9118       .0118       .9211       .0111       .9228         100       .9143       .0083       .9208       .0079       .9279         100       .9187       .0101       .9267       .0069       .9273         100       .9532       .0050       .9566       .0047       .9577         100       .9543       .0034       .9567       .0033       .9577	2	50	.8853	5610.	7706.	.0185	.9013	.011	23.0
50       .9021       .0143       .9138       .0135       .9158         100       .9054       .0099       .9135       .0095       .9145         50       .9118       .0118       .9211       .0111       .9228         100       .9143       .0083       .9208       .0079       .9216         50       .9187       .0101       .9267       .0069       .9279         100       .9209       .0072       .9264       .0069       .9273         50       .9532       .0050       .9566       .0047       .9577         100       .9543       .0034       .9567       .0033       .9571		100	.8926	.0130	.9043	,0124	.9034	600°	0°94
100       .9054       .0099       .9135       .0095       .9145         50       .9118       .0118       .9211       .0111       .9228         100       .9143       .0083       .9208       .0079       .9216         50       .9187       .0101       .9267       .0096       .9279         100       .9209       .0072       .9264       .0069       .9273         50       .9532       .0050       .9566       .0047       .9577         100       .9543       .0034       .9567       .0033       .9571	10	50	.9021	.0143	.9138	.0135	.9158	700°	0°94
50       .9118       .0118       .9211       .0111       .9228         100       .9143       .0083       .9208       .0079       .9216         50       .9187       .0101       .9267       .0096       .9279         100       .9209       .0072       .9264       .0069       .9273         50       .9532       .0050       .9566       .0047       .9577         100       .9543       .0034       .9567       .0033       .9571		100	4506.	6600.	.9135	9600.	.9145	.003	92.0
100       .9143       .0083       .9208       .0079       .9216         50       .9187       .0101       .9267       .0096       .9279         100       .9209       .0072       .9264       .0069       .9273         50       .9532       .0050       .9566       .0047       .9577         100       .9543       .0034       .9567       .0033       .9571	15	50	.9118	0118	.9211	.0111	.9228	.011	0°69
50       .9187       .0101       .9267       .0096       .9279         100       .9209       .0072       .9264       .0069       .9273         50       .9532       .0050       .9566       .0047       .9577         100       .9543       .0034       .9567       .0033       .9571		100	.9143	.0083	.9208	6200.	,9216	.010	138.
100       .9209       .0072       .9264       .0069       .9273         50       .9532       .0050       .9566       .0047       .9577         100       .9543       .0034       .9567       .0033       .9571	20	50	.9187	.0101	.9267	9600.	.9279	910.	92.0
50       .9532       .0050       .9566       .0047       .9577         100       .9543       .0034       .9567       .0033       .9571		100	.9209	.0072	.9264	6900*	.9273	.015	184.
.9543 .0034 .9567 .0033 .9571	100	50	.9532	.0050	9956.	2400.	.9577	940.	.094
		100	.9543	+600.	.9567	.0033	.9571	540.	920.

#### BIBLIOGRAPHY

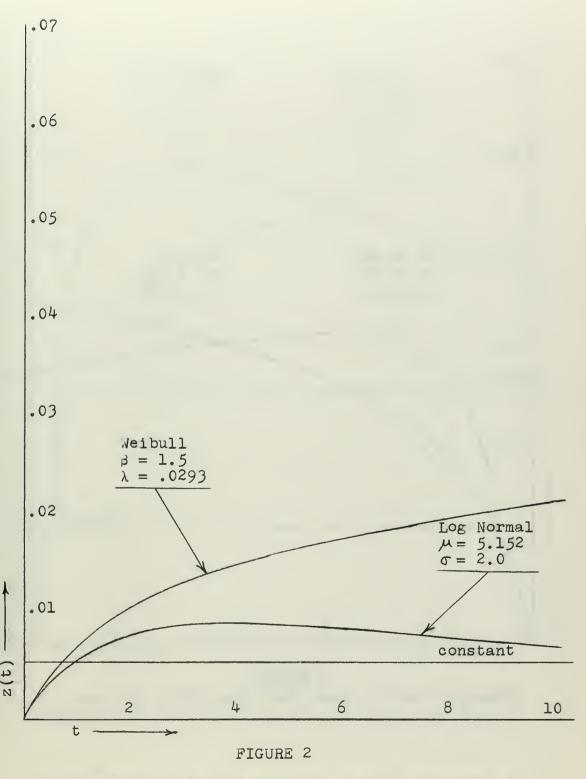
- 1. Cox, Daniel R., Renewal Theory, Methuen and Co. Ltd., London, 1962.
- 2. Guide Manual For Reliability Measurement Program (NAVWEPS OD 29304), 15 May 1965.
- 3. Meyer, Paul L., <u>Introductory Probability and Statistical Applications</u>, Addison-Wesley, Reading Mass., 1965.
- 4. Parzen, Emanuel, Modern Probability Theory and Its Applications, Wiley, New York, 1965.

APPENDIX I

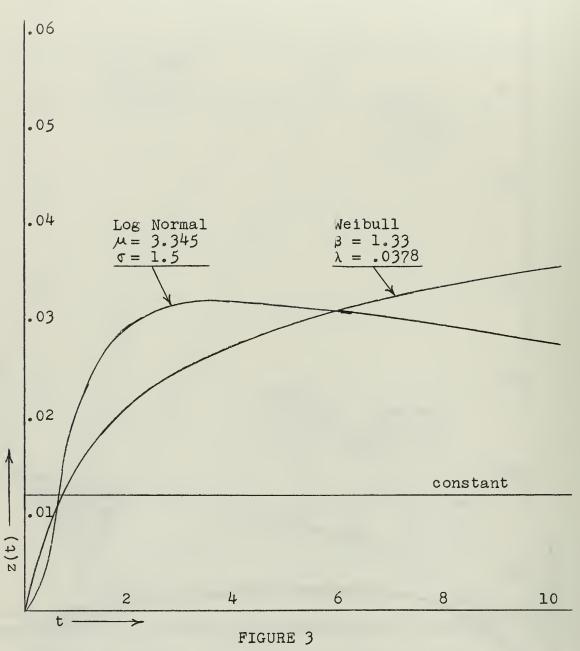
# GRAPHS OF COMPONENT FAILURE RATE FUNCTIONS FOR EACH SYSTEM TESTED



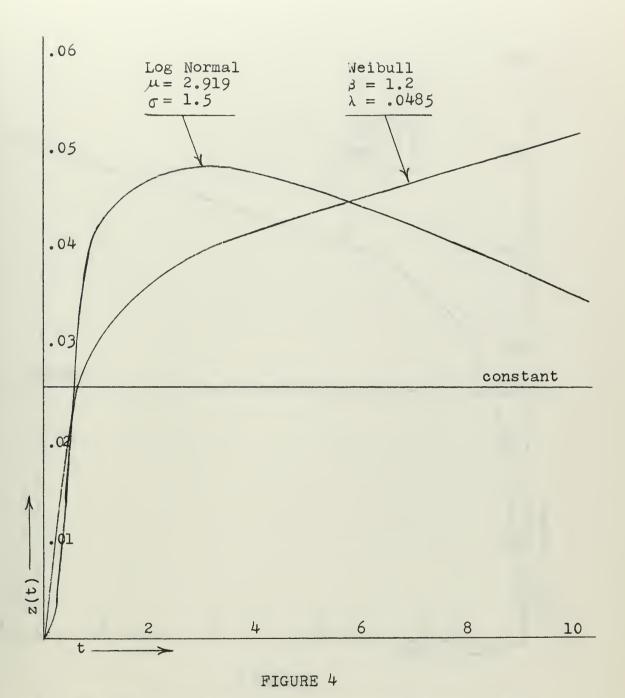
Component Failure Rate Functions, z(t), for Case 1.



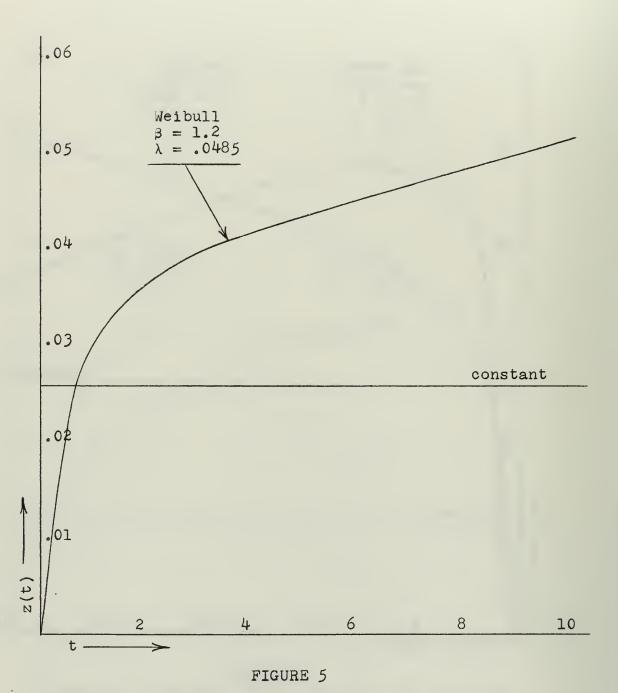
Component Failure Rate Functions, z(t), for Case 2.



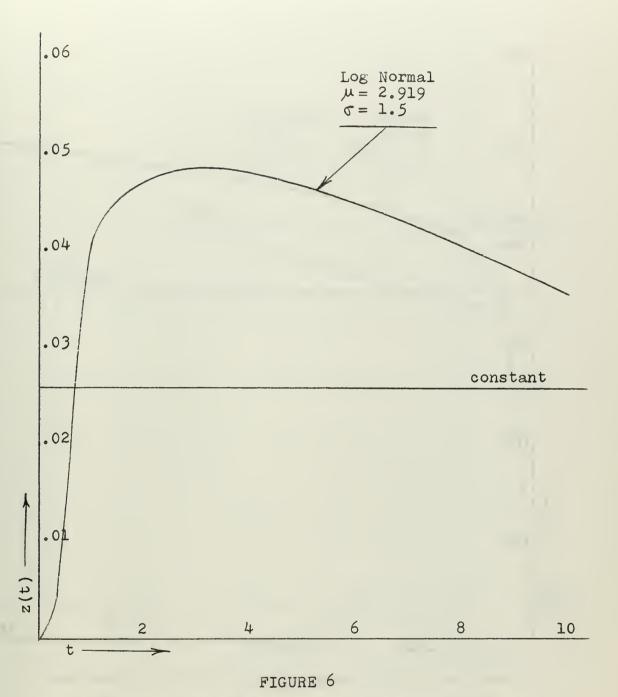
Component Failure Rate Functions, z(t), for Case 3.



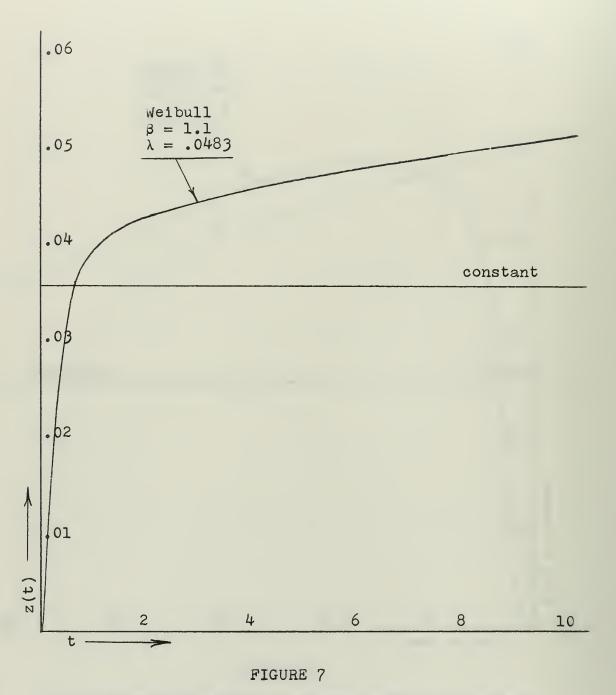
Component Failure Rate Functions, z(t), for Case 4.



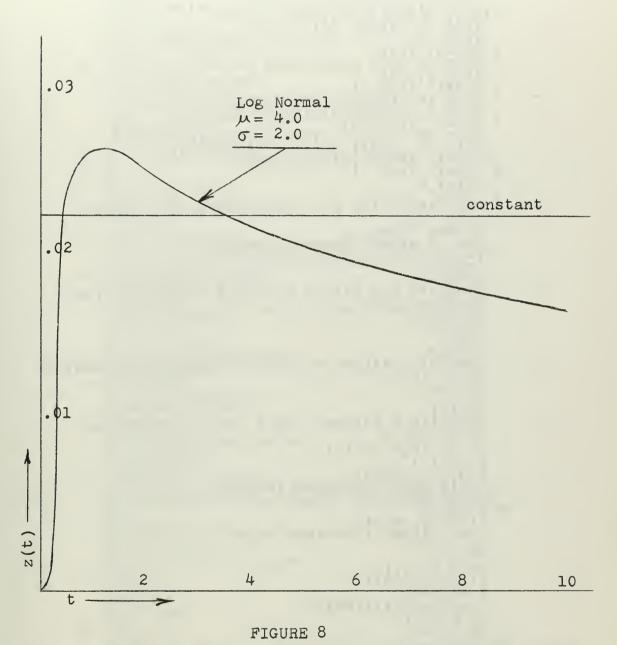
Component Failure Rate Functions, z(t), for Case 5.



Component Failure Rate Functions, z(t), for Case 6.



Component Failure Rate Functions, z(t), for Case 7.



Component Failure Rate Functions, z(t), for Case 8.

#### APPENDIX II

#### COMPUTER PROGRAM

```
DIMENSION PS(500), RSL(500), S(4), NF(4),
*N(4), HLAM(4), BTA(30), TO(4,3)
READ IN SAMFLE SIZE, N
READ 1, N
FORMAT (418)
READ IN BETA CORRECTION FACTORS
READ 2, BTA
FORMAT (5FR.3)
READ IN PLANNED TEST TIMES
READ 3, ((TO(1,J), I=1,4), J=1,3)
FORMAT (4F10.0)
READ IN FAILURE DISTRIBUTION PARAMETER
C
C
C
        READ IN FAILURE DISTRIBUTION PARAMETERS READ 12, ALFA1, BET 41, ALFA2, BET 42, *SIGMA1, XMEAN1, SIGMA2, XMEAN2

12 FORMAT (8F6.0)
C
                PRINT 4
                                4
                FORMAT (1H1)
RUN SIMULATION FOR THREE DIFFERENT PLANNED
TEST TIMES
DO 999 IRUN=1,3
URN IS RANDOM NUMBER GENERATOR
Y=URN(0)
CCC
C
                 RTOT=C.
                RLTOT=0.

GENERATE 500 VALUES OF LOWER CONFIDENCE LIMIT DO 50 K=1,500
C
                 FTOT=C.
                 TS=0.
TSUM=0.
                 SIMULATE FAILURE OF FACH OF THE FOUR COMPONENTS
C
                 S(I)=0.
NF(I)=0
                 M=N(I)
                GENERATE N FAILURE TIMES FOR EACH COMPONENT
DO 20 J=1, M
GO TO (5,6,7,8), T
~
                Y=URN(1)

RTNV1=1./RETA1

TP=((-ALOG(Y))**RINV1)/ALFA1

GO TO 10
                Y=URN(1)
PINV2=1./BET42
TP=((-ALOG(Y))**BINV2)/ALF42
50 TO 10
          GO TO 10
7 TS1=0.
00 9 IJ=1,12
S TS1=TS1+URN(1)
X1=TS1-6.
71=SIGMA1*X1+XMEAN1
TP=EXP(71)
GO TO 10
8 TS2=0.
00 11 IJ=1,12
1 TS2=TS2+URN(1)
X2=TS2-6.
                X2=TS2-6.
Z2=SIGMA2*X2+XMEAN2
TP=EXP(Z2)
CONTINUE
        19
                COUNT NUMBER OF FAILURES BEFORE PLANNED TEST TIME EXPIRES IF (TP.LT.TO(1, TRUN)) GO TO 15 T=TO(1, IRUN)
C
```

```
GO TO 18
       15
              T = TP
               NF(T)=NF(T)+1
               S(I)=S(I)+T
       18
              CONTINUE
CALCULATE LAMDA HAT AND SIMULATED SYSTEM
RELIABILITY
       ŽČ
               XN=N(I)
              F=NF(I)
HLAM(T)=(F/S(I))*(2.*XN/(2.*XN+1.))
               FTOT=FTOT+F
              HEAMD=HEAMD+HEAM(I)
TSUM =TSUM+(HEAM(I)/S(I))
TS=TS+1./S(I)
CONTINUE
              CONTINUE
RS(K)=EXP(-HLAMD)
RTOT=RTOT+RS(K)
CALCULATE UPPER LAMDA HAT AND
LOWER CONFIDENCE LIMIT RELIABILITY
IFTOT=FTOT+1.
IF(IFTOT-3!) 31,32,32
BETA=BTA(IFTOT)
       31
               60 TO 33
            GO TO 33

RETA=1.

XK=(.842)*RETA

TF(FTOT) 35,35,40

ULAMD=(XK**2)*TS/4.

GO TO 41

CHAT=TSUM/HLAMD

ULAMD=(2.*HLAMD+(XK**2)*CHAT+SQRT(4.*HLAMD*

*(XK**2)*CHAT+(XK**4)*(CHAT**2)))/2.

RSL(K)=EXP(-ULAMD)

RLTOT=RLTOT+RSL(K)

CONTINUE
       32
33
       35
       50
               CALCULATE MEANS AND STANDARD DEVIATIONS REAR = RTOT/500,
C
               SIJM=0.

DD 60 K=1,500

DIFF=(RS(K)-RPAR)**2

SUM=SUM+DIFF
              CONTINUE
SR=SORT(SUM/500.)
RLBAR=RLTOT/500.
       60
               SUM=0.

DO 61 K=1.500

DIFF=(RSL(K)-RLBAR)**2

SUM=SUM+DIFF
              CONTINUE

SRL=SORT(SUM/500.)

ORDER 500 VALUES OF LOWER CONFIDENCE LIMIT
ON RELIABILITY AND SELECT BOTH PERCENTILE.
       61
CC
               II = 1
              B=RSL(1)
DO 70 I=1,500
IF(B-RSL(I)) 65,70,70
B=RSL(I)
               00 80 K=1,101
       65
               [ ] = ]
               CONTINUE
               RSL(II)=0.
              A=B
CONTINUE
PRINT OUT VALUES OF MEANS AND STANDARD DEVIATIONS
AND SCTH PERCENTILE
PRINT 90, RLBAR, SRL, RBAR, SR, A
FORMAT(//1CX, 5F2).5)
CONTINUE
END
        29
Č
       90
     990
```

#### APPENDIX III

## WEIBULL DISTRIBUTION 6

The Weibull Probability Distribution is defined by:

$$f(t) = \beta \lambda^{\beta} t^{\beta} - 1_{e} - (\lambda t)^{\beta} \qquad 0 \le t \le \infty$$

where  $\lambda$  is the "scale" parameter, and  $\beta$  is the "shape" parameter.

The Cummulative Distribution Function (CDF) is then

$$F(t) = \int_0^t f(x)dx = 1 - e^{-(\lambda t)^{\beta}}$$

and the failure rate function is

$$z(t) = f(t)/R(t) = \lambda^{\beta}\beta t^{\beta} - 1$$

where R(t)=1-F(t)= the reliability function. The Weibull Probability Distribution reduces to the Exponential Probability Distribution when  $\beta=1$ . Increasing failure rates are obtained only when  $\beta>1$ .

<sup>&</sup>lt;sup>6</sup>Daniel R. Cox, <u>Renewal Theory</u> (London: Methuen and Co. Ltd., 1962) p. 21.

#### APPENDIX IV

### LOG NORMAL DISTRIBUTION 7

The Log Normal Probability Distribution is given by

$$f(t) = \frac{1}{\sqrt{2\pi} \sigma t} e^{-\frac{1}{2} \left[ \frac{\ln t - \mu}{\sigma} \right]^2}$$

$$= e^{X}$$

where X is distributed as Normal ( $\mu$ , $\sigma^2$ ). For computational purposes, it is beneficial to derive the Cummulative Distribution Function as

$$F(t) = P(t \le T)$$

$$= P(t \le e^{X})$$

$$= P(\ln t \le X)$$

$$= P\left[\frac{\ln t - \mu}{\sigma} \le Z\right]$$

$$= \Phi\left[\frac{\ln t - \mu}{\sigma}\right]$$

$$f(t) = F'(t) = \frac{1}{\sigma^{t}} \Phi\left[\frac{\ln t - \mu}{\sigma}\right]$$

then

The failure rate function is given by:

$$z(t) = f(t)/R(t) = \frac{\frac{1}{\sigma t} \Phi \left[ \frac{\ln t - \mu}{\sigma} \right]}{1 - \Phi \left[ \frac{\ln t - \mu}{\sigma} \right]}$$

<sup>7</sup> Paul L. Meyer, <u>Introductory Probability and Statistical Applications</u> (Reading, Mass: Addison-Wesley, 1965) p. 187.

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Phis study evaluates the accuracy of an established reliability measurement procedure (NAVNEPS OD 29304) by computer simulation. The reliability measurement procedure assumes components fail accurding to an Exponential Failure Law. This study tests the accuracy of that procedure when components obey a weibull Failure Law or a Log Normal Failure Law.

3 ABSTRACT

Security Classification									
KEY WORDS	LINK A		LINKB		LINK C				
	ROLE	WT	ROLE	WT	ROLE	WT			
Reliability									
2-71-14715-14-14									
Reliability Measurement									
Reliability Confidence Limit									
.√eibull Distribution									
Log Normal Distribution									
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